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TECHNICAL REPORT

DEVELOPMENT COMPOUNDING AND EVALUATION OF PHOSPHAZENE RUBBER FOR HELICOPTER SEAL APPLICATIONS

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Angus Wilson

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October 1974

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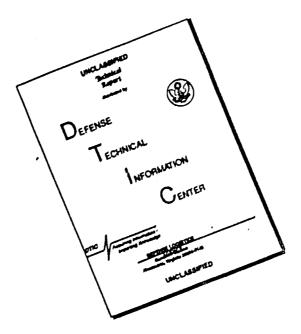
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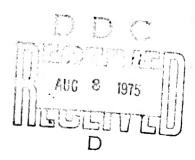
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# DEVELOPMENT COMPOUNDING AND EVALUATION OF PHOSPHAZENE RUBBER FOR HELICOPTER SEAL APPLICATIONS

by

ANGUS F. WILSON



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#### FOREWORD

This report covers the results obtained from the development compounding and testing of a polyphosphazere rubber copolymer with the structure:

$$-\text{ENP} (\text{OCH}_2\text{CF}_3)_2 - \text{NP} (\text{OCH}_2\text{C}_3\text{F}_6\text{CF}_2\text{H})_2 \frac{1}{x}$$

Objective of the work was the attainment of optimum strength, and the evaluation of physical properties and heat resistance with a view toward possible use of the material in helicopter seal

The base rubber was supplied by Horizons, Inc., a contractor of the Army Materials and Mechanics Research Center (AMMRC). Work was carried out for the Organic Materials Laboratory of AMMRC under a customer order, PRON. NO. AW-1-Y1-1735-O1-AW-BG. Funds for the project were furnished by AVSCOM under Project 1716042.

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## Development Compounding of Phosphazene Rubber for " licopter Seal Applications

#### I. Introduction

Phosphazene rubber (or phosphonitrilic fluoro-elastomer) was developed under an Army Materials and Mechanics Research Center (AMMRC), Watertown, Massachusetts contract with Herizons, Inc., Cleveland, Ohio. The original work was directed toward the development of a fuel and oil resistant rubber that would remain flexible at low temperatures. Under the program several polymers were made and one of them, a copolymer with the structure:

$$-\text{ENP}(\text{OCH}_2\text{CF}_3)_2 - \text{NP}(\text{OCH}_2\text{C}_3\text{F}_7)_2 \frac{1}{x}$$

underwent development compounding and testing at NLABS from 1968 to 1970. The data obtained showed the material to be a rubber with excellent fuel resistance, and with flexibility down to approximately -55°C, although the physical properties were poor, i.e. tensile strength below 61.2 Kg/cm<sup>2</sup>\* (870 psi). 1,2,3

In 1972 AMMRC expanded the program with Horizons, Inc. to include the development of candidate phosphazene rubber compounds for evaluation as seal material for helicopter transmission shafts. The polymer chosen for this was a copolymer with the structure:

$$-\text{ENP}(\text{OCH}_2\text{CF}_3)_2 - \text{NP}(\text{OCH}_2\text{C}_3\text{F}_6\text{CF}_2\text{H})_2\frac{1}{2x}$$

and the target properties sought were a Shore A durometer of 75 to 90, minimum tensile strength of 105.5 Kg/cm<sup>2</sup>(1500 psi), and minimum ultimate elongation of 175%. After heat aging 70 hours at 150°C the allowable maximum changes were to be +10 points durometer, -25% tensile strength, -30% ultimate elongation.

As part of the helicopter seal program AMMRC transferred funds to NLABS for development compounding to attain the target properties, and for evaluation of the materials produced. Compounding was also carried out by Horizons, Inc., suppliers of the polymer, and later in the program by personnel at the Mobility Equipment Research and Development Center at Fort Belvoir, Virginia.

\*Tensile strengths are expressed as  $Kg/cm^2$  in accordance with the ASTM method used in their measurement. If desired, the data can be converted from  $Kg/cm^2$  to pascal by multiplying the given value by 9.806 x 10.

#### II. Materials

Five batches of copolymer- $\{NP(OCH_2CF_3)_2 - NP(OCH_2C_3F_6CF_2H)_2\}_{x}$  were received from April through November, 1972 and were identified as follows:

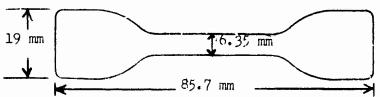
Batch	Amount, grams	Viscosity, [n], acetone
1785-47A	246	1.85
1829-20	<b>2</b> 06	2.41
1869-34	450	1.8
1785-47 <b>A</b> (2)	450	1.85
1931-07	910	1.71

The identity and source of other materials used in the compounding studies are given in the Appendix.

Tests on the raw polymer showed it to be soluble in acetone and methyl alcohol, and insoluble in benzene and 70/30 isooctane/toluene by volume.

#### III. Compounding, Curing, and Testing Procedures

Generally, batches using 5 grams of rubber were prepared. Ingredients were weighed on an analytical balance and compounded on a rubber mill with 25-mm (l-in.) diameter rolls, turning at a 10/7.5 speed ratio. After mixing, the rubber compound was removed from the 25-mm mill and sheeted to the desired thickness on a 76-mm (3-ln.) mill. It was then molded and cured using a 1.9-mm (0.075-in.) thick mold of the following configuration and dimensions:



The use of teflow sheeting on either side of the mold cavity, required when molding previous phosphazene rubbers was unnecessary with this copolymer, and a light silicone spray coating was found sufficient to provide release.

Oven post curing, where used, was done in a circulating air oven of the type specified in ASTM Method D573-67.

Tensile and elongation testing was done according to ASTM Method D412-68, except that a non-standard test specimen (shaped to the mold dimensions shown above) was used.

Shore A durometer values were obtained according to ASTM Method D-2240-68. The tensile test specimen was used for this test and values were obtained on the tab ends of doubled specimens, backed up by another rubber material of similar durometer to provide sufficient thickness.

Low temperature flexibility (Gehman torsional stiffness) values were obtained according to D-1053-65.

# IV. Results and Discussion

# A. Preliminary Compounding

In order to compare the new material with the rubber previously evaluated, a sample of batch 1785-47-A, was compounded in a recipe which with the previous material had given a tensile strength of 61.2 Kg/cm<sup>2</sup> (870 psi) - the highest obtained. The recipe in parts by weight was as follows:

Polymer	100
Hisil 233	15
Magnesium oxide	5
Dicup 40C	L

The first three ingredients were mill mixed and bin-aged one week at room temperature prior to milling in the Dicup 40C. The specimen was then press-cured 1 hour at 143°C; press-coooled; and oven-post-cured 1 hour at 93°C, 1 hour at 104°C, 1 hour at 121°C and 16 hours at 141°C; and gave the following results:

Tensile strength. 23.2 Kg/cm<sup>2</sup>(330 psi)
Ult. Elongation 200%
Shore A durometer 45

The compound was repeated but without bin-aging, and the material was mill mixed and press cured the same day, with the following results:

Tensile strength 44 Kg/cm<sup>2</sup>(625 psi)
Ult. Elongation 200%

ore A durometer 45

Although part of this higher tensile result could be attributed to experimental difference, it appeared that bin-aging had no beneficial effect on the tensile strength of this compound. Essentially the same compound was again mixed, except that 1.5 parts of stearic acid was added as a lubricant, and to remedy a tendency of the compound to stick to the small mill rolls during mixing. The addition, while resulting in a smoother sheet off the mill, did little to relieve the tendency to stick, and the cured material had many small surface bubbles. This batch also was mill mixed and press-cured the same day, and gave these results:

Tensile strength

42.2  $Kg/cm^2$  (600 psi)

Ult. Elongation

250%

Shore A durometer

46

Since the recipe used above was a basic and simple formula, compounds were mixed eliminating various ingredients to determine their effect on physical properties. A compound was mixed containing only the polymer and dicumyl peroxide as follows:

Polymer

100

Dicup 400

4

Press cured

60 min/127°C

Oven cured, for:  $1 \text{ Hr/93}^{\circ}\text{C}$ ,  $1 \text{ Hr/104}^{\circ}\text{C}$ ,  $1 \text{ Hr/121}^{\circ}\text{C}$ , and then  $16 \text{ Hr/141}^{\circ}\text{C}$ 

This gave a tensile strength of less than 0.7 Kg/cm<sup>2</sup>(10 psi), comparable to that of a specimen of press-molded gum rubber.

To test the effect of magnesium oxide, the compound was again mixed, this time with the addition of 5 parts of the oxide, and presscured and oven cured as before. The results this time were 17.6 kg/cm<sup>2</sup>(250 psi) tensile strength, and 330% ultimate elongation. Thus, this initial series of compounds shows that dicumyl peroxide alone causes little, if any, effective curing of the phosphazene copolymer. Addition of magnesium oxide to the peroxide causes a marked improvement in the tensile strength, and the addition of a silica filter to these two ingredients increases the tensile again, by a factor of 2.5.

#### B. Fractionation

To check the possibility that the tensile strength of the rubber could be improved by removing its lower molecular apportion, a small quantity of the raw gum was fractionated by

coagulation from an alcohol solution: 25 grams of raw polymer were dissolved in 200 ml. of reagent methanol, and 300 ml. of benzene was added dropwise, with stirring, over a period of 1½ hours. The coagulated polymer was allowed to settle out and the liquid decanted. The polymer was air-dried at room temperature for 18 hours, and at 85° to 88°C for one hour. The yield was 80%, i.e. 20 grams, and the material was less tacky, and appeared somewhat tougher, than the parent polymer. The decanted liquid was evaporated at room temperature and the residue dried for one hour at 93°C. This material was flowable, soft and very tacky, with apparent low viscosity.

The coagulated rubber portion was cured according to the following recipes, with the following results:

Fractionated Polymer	100	100	100
Hisil 233	15	25	15
Magnesium oxide	5	5	<u> </u>
Dicup 40C	4	2.5	2.5
Press cure Min/°C	60/143	60/143	60/143
Oven cure Hrs/OC	1/93	1/93	1/93
	1/104	1/104	1/104
	1/121	1/121	1/121
	16/141	16/141	16/141
Tensile, Kg/cm <sup>2</sup> (psi)	47.1(670)	47.1(670)	40.8(580)
Ult. Elong., %	180	1.0	180
Shore A duro.	45	67	46

The data, when compared with that obtained on whole polymer, compounded in similar recipes (see above), shows no real superiority for the fractionated material, and would indicate that the presence of low viscosity material was not, or at least was not alone, responsible for the low strength values obtained on this copolymer.

#### C. Heat Treatment

The effect of heat applied during various phases of compounding was investigated using the following recipe:

Polymer 1785-47A	1.00
Maghite D	5
H sil 235	18
A 172*	1.
Diegon ACC	),

The hompound was press-rured 1.3 hours at 143°C and oven cured 45 minutes at 93°C, 45 at 104°C, 45 at 121°C, and 16 hours at 143°C, and gave the following values:

Tensille strength	38.7 Kg/cm <sup>2</sup> (550 psi)
Ult. Elongation	280%
Shore A durometer	50

The compound was repeated and all the ingredients except the dicumyl peroxide were milled together and oven heated for 16 hours at 100°C. It was then returned to the mill for addition of the peroxide and was cured as above, with these results:

Temsile strength	56.3 Kg/cm <sup>2</sup> (800	psi)
Ult. Elongation	260%	
Shore A Suremaker	<b>51</b>	

The original compound was again repeated but no oven cure was used, and these values were obtained:

Tensile strength	66.8 Kg/cm <sup>2</sup> (950 psi)
Vitt. Elongation	280%
Shore a durometer	<i>L</i> 3

Another mix of the compound was made, in which all ingredients except the peroxide were mill mixed and oven heated for 16 hours at 100°C, prior to final milling, and which was not oven-postcured gave the following:

List Minus 114		di	WILLCIT	was	1100	CASU-TO!		_	
Ten	sile st	rengl	in			70.3	$Kg/cm^2$	(1000	psi)
Ult	. Elong	ation	ì			260%			
Sho	re A du	romet	ter•			40			

The series shows that mill mixing and oven heating of ingredients before final mixing and curing improves physical properties, and that elimination of oven-posteure also results in improved properties. However, the combination of preheating and oven cure elimination gave only slightly higher values

\*A 172 is an organo silane (see Appendix), one of a class of materials currently finding use in rubber compounding to promote interaction between rubber and non-black filters (5).

than elimination of oven cure alone, and the difference, in fact, can be attributed to normal test variation. The results strongly indicate that lengthy oven-postcuring is unnecessary with the dicumyl peroxide cure system.

There was a possibility that the heat resistance of the cured rubber could be improved by heating the raw rubber prior to compounding, on the hypothesis that such treatment would eliminate those portions of polymer most heat susceptible and leave a residue of heat resistant material. Accordingly a 15-gram sample of raw rubber was heated for 96 hours at 149°C. The material lost 0.5% of its weight and became dark, transparent, and flowed out flat and level. It was compounded using the recipe given above and gave a soft putty-like mass, which when presscured for 60 minutes at 143°C gave a tensile of less than 0.7 kg/cm²(10 psi).

A second sample of raw rubber was mill mixed with the Hisil 233 (from the above recipe) and then heated for 96 hours at  $149^{\circ}$ C. This material lost 1% of its weight but did not darken to the extent that the raw rubber had, and did not flow, or lose its shape. It was compounded with the remaining recipe ingredients and press-cured, and gave a tensile strength of less than 0.7 Kg/cm<sup>2</sup>(10 psi).

These mixes clearly show that the heating at 149°C in air results in severe degradation of the base polymer, probably due to depolymerization. Allcock, Kugel and Valan had demonstrated that methoxy, ethoxy, trifluoroethoxy and phenoxy derivatives of (NPCl<sub>2</sub>)<sub>n</sub> suffered depolymerization below 200°C, and that solution viscosity measurements showed a viscosity decrease after heating 8 hours at 150°C. Further, their work showed retention of form and integrity when treated in a vacuum at 200°C, but depolymerization in mitrogen at 150°C and 200°C.

#### D. Solution Mixing

o check the efficiency of the milling procedure, especially in the mixing of the filler into the base rubber, a compound was made by dissolving the rubber and then adding the filler; as follows:

Twelve grams of polymer 1785-1.7A were dissolved in 100 ml of methanol, and 2.16 gms of Hisil 233 (equivalent to 18 parts per hundred parts of rubber) were stirred into the solution. The mixture was allowed to stand for three

days and then poured into an evaporating dish with three methanol rinsings of 15 m) each. The methanol was allowed to evaporate away at room temperature for one day. The polymer/filler residue was dried for 2 hours at 80°C, will mixed with magnesium oxide and dicumyl peroxide, and press and oven cured. The results obtained were:

Thensile strength

 $72.1 \text{ Kg/cm}^2 (1025 \text{ psi})$ 

Ult. Elongation

310,8

Shove A durometer

36

These results were comparable to the results obtained on mill mixed stocks discussed in the previous section and showed mixing on the 1" mill to be an efficient mixing procedure.

#### E. General Compounding

series of apounds in which both rubbers were formulated in the same recipes. This, in effect, gave a duplicate test of the individual recipes and provided an opportunity to observe any trend toward a difference in the properties of the two base rubbers. The compounds formulated and the results obtained, listed in Table I, show that in most instances there was no significant difference between the two batches of polymer. The sole exception was the pair 47-A-10 and 20-10 made with filler of silica and carbon black, with which the 1829-20 polymer gave a tensile strength of 70 kg/cm<sup>2</sup> (995 psi), compared to 14 kg/cm<sup>2</sup> (200 psi) for the 1785 - 47A polymer.

Double batches of these two compounds were prepared, split in half, and press-cured. One specimen of each was oven-postcured and the other was not. Compound 1829-20 gave a tensile of 67 Kg/cm<sup>2</sup> (955 psi) with oven cure and 56 Kg/cm<sup>2</sup> (795 psi) without; 1785-47A gave 46 Kg/cm<sup>2</sup> (655 psi) with oven cure and 39 Kg/cm<sup>2</sup> (555 psi) without. Although the 1829-20 polymer again gave the higher strength results, the difference was not as great as with the previous mix. Another mix of the compounds was prepared, this time with results of 67 and 26 Kg/cm<sup>2</sup> (955 and 355 psi) respectively. The two compounds were then mixed with a,a'-bis(t-butyl peroxy) diisopropylbename (Vulcup 40KE) substituted for the dicumyl peroxide with similar results of 65 and 17 Kg/cm<sup>2</sup> (925 and 245 psi). Thus the results show the tensile values obtained with polymer 1829-20 to be consistent, while those with polymer 1785-47A to be erratio, indicating a sensitivity of this material to the silica/carbon black filler in conjunction with peroxide cures. The basic reason for this is not apparent.

Otherwise, a review of the results contained in Table I (see Appendix B) shows that the highest strength values were obtained with the silica filler, dicumyl peroxide cure (54.8 Kg/cm<sup>2</sup>) (780 psi) and with a HAF black filler, dicumyl peroxide cure (59 Kg/cm<sup>2</sup>) (840 psi), values still short of the target value of 105 Kg/cm<sup>2</sup> (1495 psi).

The use of phenol-formaldehyde resins (Amberol and Durez) in conjunction with either perceide or amine cures gave poor tensile properties. Likewise, the use of the Diaks, developed for curing fluoroelastomers, resulted in complete degradation of polymer strength properties. This could have been due either to the effect of materials themselves, the higher temperatures required for their activation, or a combination of these factors.

In order to conserve polymer, since the amounts available were relatively small, the practice of making a group of compounds from two different batches of polymer was discontinued. Instead a series of compounds was made to evaluate the effects of various curing agents, fillers, coupling agents, and other additives. The recipes used and the physical properties obtained are given in Table II.

Analysis of the data allows several conclusions to be drawn.

The use of a base, such as TETA or 2,6-diaminopyridine in conjunction with dicumyl peroxide (47-A-12, 57-A-16, 34-K) resulted in tensile reductions of up to 50%.

A silica/HAF black filler combination (47-A-17) gave mediocre physical properties.

The blending of polyphosphazene polymer with ethylene propylene diene rubber (20-11, 20-12) gave tensile properties inferior to those of straight polyphosphazene.

The use of dibasic lead phosphite (Dyphos) in a silica filled and dicumyl peroxide cured stock (20-13, 20-14) slightly improved tensile properties.

A group of compounds made to investigate dicumyl peroxide levels and cure times (34-A thru 34-H) showed that 2 parts of Dicup 400 gave tensile properties equivalent to, or slightly higher than, those obtained with 4 parts, and that better properties were attained without oven postcure than with it. This is in agreement with results contained in Table I, previously discussed.

tondablens in percented levels in conjunction with the use of dibasic lead phosphile; in compounds subjected to oven heating prior to peroxide addition (3k-K, 3k-K, 34-K Rpt and 34-J Rpt) showed no difference between 2 and 4 names of Disup 400, and no tensile improvement resulting from oven preheating. However, the dibasic lead phosphite, sold commercially as a light and heat stabilizer for bihorinated rubbers and vinyle, may disting the effects normally resulting from heat.

Compounding without any filler other than 20 parts of dibasic lead phosphite (35-L) gave a very low tensile.

Compounding with a high level of silica (50 parts) with and without the coagent Cheminic 30 (34-M, 34-N) gave a stiff material incapable of being best without breaking.

Vulcup 40KE (40% a-a'bis(t-butyl peroxy) diisopropylbenzene on clay) in conjunction with varying amounts of silica (34-0, 34-P, 34-Q) gave tensiles up to 91.4 Kg/cm<sup>2</sup> (1300 psi), but in the presence of 2,6-diamino-pyridine (34-R) gave poor results. The use of a coagent, Chemlink 30, with the Valeup (34-S) did not improve the strength.

The use of a clay filler, such as Burgess OPE (34-T through 34-W) resulted in provise medicare cures. The use of a silane treated clay, Burgess RE, alone or in combination with a silica (34-AN, 34-AO) showed substantial improvement, 56 to 74 kg/cm<sup>2</sup> (800 to 1055 psi) compared to 15 to 44 kg/cm<sup>2</sup> (215 to 625 psi).

A series of compounds using a variety of clay, black and silica fillers (34-X, 34-Y, 34-Z, 34-AA through 34-AP, 34-AT through 34-AZ, and 34-BA through 34-BH), in combination with Vulcup curing agent ranged in tensile strength from 21 to 116 Kg/cm (300 to 1650 psi). Those meeting the target tensile of 105.5 Kg/cm<sup>2</sup> (1500 psi) were compounds using a 18/40 combination of silica and silane treated clay (34-AC), a silane treated silica 34-AH, and combinations of silane treated clay and silane treated silica (34-AI, 34-AJ, 34-AK). In all of the batches additional silane was added to the fillers prior to milling. The highest tensile strength achieved was in compound 34-AI with a value of 116.0 Kg/cm<sup>2</sup> (1650 psi), evidently because of the combination of clay, silica, and silanes used in the filler system.

Three additional peroxides; 2,5-dimethyl-2,5-bis(t-butyl peroxy) hexane; 2-5-dimethyl-2,5-bis(t-butyl peroxy) hexyne-3; and di-t-butyl peroxide (34-AQ, 34-AR, 34-AS) all failed to achieve the tensile strength achieved with the Vulcup 40KE.

Two additional batches of raw rubber evaluated in Vulcup cured stocks using silica/clay filler combinations (07-A, 07-B, 47-A through 47-D) gave slightly lower tensile values than those obtained with the previously evaluated rubber (34-AI), i.e. 98.4 and 102.0 Kg/cm<sup>2</sup> (1400 & 1450 psi) compared with 116 Kg/cm<sup>2</sup> (1650 psi).

A series of compounds containing varying amounts of silane coupling agents in conjunction with a fumed silica filler (Aerosil 200) and cured with Vulcup (47-G through 47-V) showed no advantage over the base compound made without silane (47-F). Of the four silanes used: A151, vinyltriethoxy silane; A171, vinyltrimethoxy silane; A172, vinyltriethoxy silane; (beta-methoxyethoxy) silane; and A1100, gamma-aminopropyltriethoxy silane; 2 parts of A1100 gave the highest tensile value, 91.4 Kg/cm<sup>2</sup>(1300 psi); but this also was attained with no silane and the results indicate that no beneficial effect is obtained from addition of silane to this filler, in this polymer.

Use of alumina filler or alumina/silica blends (47-Alon, 47-Alonsil A, 47-Alonsil B) gave mediocre strength values. 81  $Kg/cm^2$  (1150 psi) or less.

The use of hydrophillic precipitated silicas (47-W through 47-AB), or naturally occurring amorphous silicas (47-AE through 47-AH) resulted in low strength values.

As previously discussed the highest tensile achieved in the Table II compounds was 116 Kg/cm<sup>2</sup> (1650 psi). However, Horizons, Inc., producers of the various batches of polymer had achieved tensile values in the 140 Kg/cm<sup>2</sup> (2000 psi) range, and with one compound had obtained 150.8 Kg/cm<sup>2</sup> (2145 psi). In order to evaluate the consistency of properties between different batches of raw rubber, Horizons' recipe was used with polymer 1785-47A (47-AC and 47-AD). Samples of the individual compounding ingredients were obtained from Horizons and used to mix the compound and, except for the peroxide, were either oven dried, milled together and bin aged; or vacuum oven dried, milled together and bin aged prior to peroxide addition and cure— the latter procedure being one followed by Horizons. The highest tensile obtained was 80.9 Kg/cm<sup>2</sup> (1150 psi). A similar series prepared

using Vulcup 40KE, instead of the Luperox 500 40KE provided by Horizons, gave a high tensile of 91.4 Kg/cm<sup>2</sup> (1300 psi). The difference of 70 Kg/cm<sup>2</sup> (1000 psi) between Horizons' and our compounding of the same recipe is strong indication of either basic strength differences in various batches of the polymer or strong sensitivity of the polymer to differences in compound mixing procedures.

The effects of various heat stabilizers- Dyphos (dibasic lead phosphite), Dythal XL (dibasic lead phtalate), and Leadstar (normal lead stearate) - in prevention of deterioration during cure were investigated in compounds 47XXO through 47XX3 and 47ZZO through 47ZZ3. Dyphos slightly improved the tensile strength, Dythal XL slightly decreased it, and Leadstar resulted in approximately a 50% reduction. One part of SRF black was also evaluated (47XX4 and 47ZZ4) because work by J. A. Williams showed that a small amount of black acts as a heat stabilizer in silicone rubber. In these polyphosphazene rubbers, however, it had lattle effect.

with polymer batch No. 1931-07. These include various black and non-black fillers, blends with another polymer, and the use of various compounding additives - all compounds being cured with Vulcup 40KE. The series again clearly demonstrates that little tensile strength can be developed without a reinforcing filler, and that the best tensile values could be achieved using the fumed silica or a silica/silane-treated-clay combination. The highest value obtained was 98.4 Kg/cm<sup>2</sup> (1400 psi), using such a combination. The results clearly demonstrated that carbon black fillers generally gave very poor results and in many cases prevented curing. The best tensile obtained with a black was 63.3 Kg/cm<sup>2</sup> (900 psi), and this was in a compound using a blend of polyphosphazene rubber and EPDM rubber (02-G). Blending of the two rubbers resulted in low tensile values, the above value being the highest obtained.

The series also showed that a satisfactory cure could be obtained by substituting zinc oxide for magnesium oxide, and that the use of ultrafine talc particles as a filler produced stocks with moderate tensile strength; i.e. approximately 49 Kg/cm<sup>2</sup> (700 psi).

#### F. High Temperature Aging and Seal Testing

In order to evaluate the ability of the polyphosphazene rubber to meet the target requirements for resistance to heat aging, compound 34-AC (Table II) was mixed, press cured 20 minutes at  $160^{\circ}$ C and oven cured 18 hours at  $100^{\circ}$ C. Physical properties were measured before and after aging 70 hours at  $149^{\circ}$ C, with the following results:

	Original	Aged	Change	Target Change
Tensile, Kg/cm <sup>2</sup>	112.5	91.4	-19%	-25% Max.
Elong., %	80	75	- 6%	-30% Max.
Shore A, Duro. Points	85	88	+ 3	+10 Max.

The tests showed good retention of properties, under these aging conditions, and as a result a 50-gram batch of the compound was milled without peroxide for fabrication into test seals. A small portion of the compound was sampled at NLABS, milled with the peroxide curing agent, cured and tested, with these results:

Tensile: 112.5 Kg/cm<sup>2</sup> (1600 psi)

Elongation: 90%

Shore A: 79

The remainder of the compound and curing agent was sent to Federal-Mogul, Inc. a contractor engaged in testing prototype helicopter seal materials supplied by Horizons, Inc. The curative was milled into the compound and seals were fabricated by Federal-Mogul for evaluation on their test equipment.

As reported by Federal-Mogul<sup>8</sup>, the test conditions were: MIL-L-7808F, 230°F, 5500 RPM, 20 hours on, 4 off, test to be terminated when measurable leakage (of the lubricant) occurred; with these results:

Seal 1: dry at 18.0 hours, leaked 16 grams after 42 hours.

Seal 2: dry at 17.5 hours, leaked 5.9 grams after 89.5 hours.

These values were intermediate between those obtained by Federal-Mogul on other polyphosphazene compounds.

#### G. Low Temperature Properties

A compound containing 15 parts of silica, 5 parts of magnesium oxide, and 4 parts of Dicup 40C (which, when press and oven cured, gave a tensile strength of 23.2 Kg/cm<sup>2</sup> and a Shore durometer value of 45) was tested for low

temperature stiffness according to ASTM Method D 1053-65, with the following results:

$$T_2$$
 -47.5°C  
 $T_5$  -54.5  
 $T_{10}$  -57.5  
 $T_{100}$  -66.0

# V. Summary and Conclusions

The copolymer,  $\{NP (OCH_2CF_3)_2 - NP (OCH_2C_3F_6CF_2H)_2\}_x$  is curable with various peroxides, and the best results were obtained with dicumyl peroxide (39.5 - 41.5% active supported calcium carbonate) and a a-a'-bis(t-butyl peroxy) diisopropylbenzene (39.5 - 41.5% active supported on silane treated clay).

The best reinforcing fillers for the copolymer were precipitated and fumed silicas, silane treated clays and combinations of these. Carbon blacks in general gave poor results and in some cases interfered with peroxide cures.

The cured rubber exhibited good aging resistance and low temperature flexibility down to  $-54\,^{\circ}\mathrm{C}_{\circ}$ 

The target properties were exceeded, and tensile strengths of 116 Kg/cm<sup>2</sup> (1650 psi) were achieved, and with the use of proper heat stabilizers higher strengths might be obtained.

# VI. Future Work

Work will continue at NLABS on compounding of these and other polyphosphazene rubbers, under an AMMRC customer order. Emphasis will be placed on achieving optimum physical properties and on developing compounds suitable for specific fabrication techniques, including dipping and coating.

#### VII. References

- (1) Wilson, A. <u>Initial Compounding Studies of Polyphosphazene Rubber</u>. TR 70-10-CE, (C&PLSEL-68) US Army Matick Labs., August 1969.
- (2) Wilson, A. <u>Polyphosphazene Rubber, Horizons, Inc., No. 1436-16,</u> 1436-19, and 1438-05. Memo Report, US Army Natick Labs., November 1969.
- (3) Wilson, A. <u>Polyphosphazene Rubber; Horizons; Inc., No. 1487-12</u>. Memo Report, US Army Natick Labs., April 1970.
- (4) Reynard, R. A., Sicka, R. W., Vicic, J. C., Rose, S. H., <u>Development of Thermally Stable Poly(Fluoroalkoxyphosphazene) for the UH-1</u>
  <u>Helicopter</u>, Final Report Contract DAAG 46-72-C-0073, September 1973
- (5) Ziemianski, L. P., Pagano, C. A. and Ranney, M. W. <u>Silanes in</u>

  <u>Flastomers: New Route to High Performance Pigmented Products.</u> Rubber World,
  October 1970.
- (6) Allcock, H. R., Kugel, R. L., Valan, K. J. <u>Phosphonitrilic</u> Compounds, VI. <u>High Molecular Weight Poly(alkoxy and aryloxyphosphazenes)</u>. Inorganic Chemistry, Pg 1709-1715, Vol. 5, No., 10, October 1966.
- (7) Williams, John A. <u>Carbon Black as a Heat Stablizer in Silicone</u>
  Rubber Vulcanizates, SWERR-TR-72-28, Weapons Laboratory at Rock Island,
  US Army Weapons Command, April 1972.
- (8) Inter-Office Memorandum, R. Hinderer to J. Born, Federal-Mogul, May 24, 1973.

# Appendix A of Compounding Ingredients

daterial	Composition	Source
Aerosil 200	Hydrophillic fumed silica	Deguary To
Aerosil 300	Hydrophillic fumed silica	Deguasa, Inc.
Aerosil R972	Fumed silica treated with	Degussa, Inc.
	dimethyldichloro silane	Degussa, Inc.
Agerite Resin D	Polymerized 1,2-dihydro-2,2,4 trimethylquinoline	R.T. Vanderbilt Co., Inc.
Alban	Funed Alumina	Cabat Camanas II
Amberol ST-137X	Phenol-formaldehyde resin	Cabot Corporation Rohm & Haas Co.
Burgess KE	Silane treated aluminum silicate	Burgess Pigment Co.
Burgess OPE	Aluminum silicate	Burgess Pigment Co.
Cebbowax 4000	Polyethylene-glycol	Union Carbide, Corp.
Chemlink 30	Trimethylol Propane Trimethacrylate	Ware Chemical Corp.
Diak 5	Treated hydroquinone	E.I. duPont deNemours & Co.
Diak 6	Proprietary, fluorocarbon curing agent	E.I. duPont deNemours & Co.
Dicup 40C	40% dicumyl peroxide on calcium carbonate	Hercules, Inc.
Durez 12687	Phenol-formaldshyde resin	Hooker Chamical Corp.
Duphos	Dibasic lead phosphite	National Lead Co.
Dythal SL	Dibasic lead phthalate	National Lead Co.
Elastomag 170	Magnesium oxide	Morton Chemical Co.
Hisil 233	Precipitated hydrated silica	PPG Industries
Imsil A10	Amorphous silica	Illinois Minerals Co.
Imsil Al5	Amorphous silica	Illinois Minerals Co.
Leadstar	Normal lead stearate	National Lead Co.
Luperco 101 XL	2, -Dimethyl-2,5-bis(t-butyl peroxy) nexane	Penwalt, Lucidol Div.
Luperco 130 XL	2,5-dimethyl-2,5-bis(t-butyl peroxy) hexyne-3	Penwalt, Lucidol Div.
Maglite D	Magnesium oxide	Merck & Co., Inc.
Mistron Vapor	Magnesium silicate	Cyprus Mines, United Sierra Div
Nordel 1040	Ethylens-propylene-diene terpolymer	E.I. duPont deNemous & Co.
Nulok 321 SP	Silane treated aluminum silicate	J.M. Huber Corp.
QUSO F22	Precipitated silica	Philadelphia Quartz Co.
QU30 G32	Precipitated silica	Philadelphia Quartz Co.
CUSO H40	Precipitated silica	Philadelphia Quartz Co.
QUSO WR82	Precipitated silica	Philadelphia Quartz Co.
Rcyalene 400	Ethylene-propylene-diene terpolymer	Unircyal Chemical Div.
Silane Al51	Vinyltriethoxy silane	Union Carbide Corp.
Silane Al71	Vinyltrimethoxy silane	Union Carbide Corp.
Silane Al72	Vinyl silane	Union Carbide Corp.
Silane Alloo	Gamma-aminopropyltriethoxy silane	Union Carbide Corn.
Silanox 101	Funed silica	Cabot Corp.
Silene EF	Precipitated hydrated calcium silicate	PPG Inudstries, Inc.
Vulcup 40KE	40% a-a'bis(t-butyl peroxy) di- isopropylbenzene on Burgess KE clay	Hercules, Inc.

## Appendix B

# Tables of Compounding Ingredients and Test Results

- Table I Duplicate Compounds of Base Polymers 1785-47A and 1829-20.
- Table II Compounds of Base Polymers 1785-47A, 1829-20, and 1869-34.
- Table III Compoun of Base Polymer 1931-07.

TABLE I

Duplicate Compounds of Base Polymers 1785-47A and 1829-20 (parts by Weight, and test results)

Polymer 1829-20         100         100         50         100	100   100   50   100	Polymer 1785-47A		1 8 1 8	1-02	100	100 100	<b>4</b> 20	100		20-5 44 A4 20-4 4(A2 20-5)	<b>†</b>	2 3 3 3	î Ş	100 100	100 100 100	100	1	100 100	001 100 100	<u> </u>	<u> </u>	3	(A) 100 100 100 100 100 100 100 100 100 10
e         5	5       5	Polymer 1829-2	0		8		81	50		18		100		8		81		158		18		1	8	100
e         15 </td <td>15 15 15 7.5 15 15 15 15 15 15 15 15 15 15 15 15 15</td> <td>Maglite D</td> <td></td> <td>2</td> <td>5</td> <td>5</td> <td>5</td> <td>5</td> <td>5</td> <td>5</td> <td>41</td> <td>5</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>c.</td> <td>67</td> <td></td> <td></td> <td></td> <td>47</td>	15 15 15 7.5 15 15 15 15 15 15 15 15 15 15 15 15 15	Maglite D		2	5	5	5	5	5	5	41	5							c.	67				47
15   15   15   15   15   15   15   15	15 15 15 17 17 18 18 18 18 18 18 18 18 18 18 18 18 18	Zinc Oxide											5	5	5	5	5	5			5		5	5
15         15<	15 15 15 7.5 1 15 15 15 15 15 15 15 15 15 15 15 15	CaOH																	9	9				
T 137X         T 15 15         T 15 15 <th< td=""><td>4       4</td><td>Hisil 233</td><td></td><td>15</td><td>15</td><td></td><td></td><td>7.5</td><td></td><td></td><td>15</td><td>15</td><td>15</td><td>15</td><td>15</td><td>15</td><td>15</td><td>15</td><td></td><td></td><td></td><td></td><td>+</td><td>01</td></th<>	4       4	Hisil 233		15	15			7.5			15	15	15	15	15	15	15	15					+	01
I 137X       I 137X       I 137X       I 15 15       I 15 I I I I I I I I I I I I I I I I I I	4       4	HAP Black				15	15	7.5													25	25		15
F137X       F13XX       F13XX <th< td=""><td>4 4 4 4 4 4 4 4 6 6 7127 60/127 - 5 7 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1</td><td>MT Black</td><td></td><td><b> </b></td><td></td><td></td><td></td><td></td><td>15</td><td>15</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td>15</td><td>15</td><td></td><td></td><td></td><td></td></th<>	4 4 4 4 4 4 4 4 6 6 7127 60/127 - 5 7 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	MT Black		<b> </b>					15	15									15	15				
87 4 4 4 4 4 4 4 4 4 6 6 6 6 6 7 8 8 8 8 8 8 8 8 8 8 8 8 8 8	4       4	Amberol ST 1	37X								<b>t</b> 0	₩	₩	٣								-		
4 4 4 4 4 4 6 4 6 6 6 6 6 6 6 6 6 6 6 6	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Durez 12687													80	æ							•	
0.8	60/142	Dicup 40XC		7	7	77	7	7	77	7	7	7									7	7	1	17
1.1	60/142 \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	Diak 5																	8	İ			i	
	60/143 ->	Diak 6																	1.1	1.1				
3 3 3 3 3 3	60/143 + + + + + + + + + + + + + + + + + + +	Triethylene tetramine											63	3	6	3	3	3					1	
$60/143 \rightarrow \qquad \rightarrow $		press cure,	ndin/℃	50/14.	£	1			1	1		$\uparrow$	60/127	, 60/12	1	1	1	1	50/173	1 1	60/11	1		

TABLE I (Continued)

Duplicate Compounds of Base Polymers 1785-47A and 1829-20 (parts by Weight, and test results)

		- 1	l i	. 1	1	l ====================================
1 1	<b></b>	*	70.3	280	3	
		+	<u> </u>			
1 1	$\uparrow$	1	14.1	330	7	
75/10,	1	7	10.5	8,	67	
75/37 + 175/104 + 1	.75/22	<u>^</u> _[	7.0 1	320	271	
75/93		16/14	7.		23 1	
1 1	1	1	0	during 1100	2	
.75/93 75/113	75/135	16/157	Broke	lurin	cure	
			8.7 E	8	۶ ک	
75/93	17	16/141 →	_		7	
75.	75/121	16/1	7.1	200	35	
1 1	1		28.1	110	077	
140	16/143→		10.5	200	07	
4 <u>5 5</u>		-			35	
1 1	$\uparrow$	1	10.5	\$000		
1 1		1	2.8	007	07	
	<b>T</b>	$\downarrow$	3.5	320	38	
	+	$\dagger$	3.5	300	07	
<del>1</del>	+	7				
1 1	<u> </u>	+	21.1	180	26	
1 1	1	+	35.1	300	23	
1 2	+	$\dagger$	31.6	210	30	
<u> </u>	+	7			<u> </u>	
1 1	$\downarrow$	1	58.6	280	23	
	$\downarrow$	+	12.2	8	8	
75/93	士	1	7.67	180	57	
1 66	-21	₹67		<b>-</b>	877	
.75/93	.75/121	16/143	54.8	250	17	
			, -		Ą	
Gven cure, hr/°C			Tensile, Kg/cm2	×	Shore A Durometer	
The,			3, 176	long	1 Dut	
5 <b>5</b>			1sile	Ult. Klong, %	ar.	
`			Ter	UŢ,	₽.	

TABLE II

Compounds of Base Polymers 1785-47A, 1829-20, and 1869-34 (parts by weight and test results)

Batch Number	47A12	47A16	47A12 47A16 47A17 20-11	20-11	30-12	20-13	20-14	34-A	34-B		<u></u>	34-E	34-F	E 5-45	H-17	34-I	7-7-	30-12   20-13   20-14   34-8   34-8   34-6   34-5   34-5   34-6   34-7   34-1   34-1   34-2   Rpt-34-1   Rpt-34-5	Rpt - 34-J
Polymer 1785-47A	18	8	100																
Polymer 1829-20				*02	81	81	18												
Polymer 1869-31								81	81	100	8	8	81	100	100	001	8	100+	100
Nordel 1040				30*		<u> </u>				<del>                                     </del>									
Maglite D	5	5	5*	2*	5	5	5	15	5	72	2	5	5	25	5	5	5	54	€5
Dyphos							5									5	5	5*	* 5
HAF Black			15*																
Hisil 233	18	18	10*	\$02	18	18	18	18	18	18	18	18	18	1.8	18	18	81	18*	18*
Сагромах 4000	0.5																		
Silane AllOO				1.1*	-	1.2	1.2												
Silane A172	1	1	1*													Ţ	1	1.	1*
Triethylene tetramine		H			i														
Dicup 400	77	77	7	7	7	7	77	7	77	77	77	2	2	2	2	7	2	7	2
Press cure, min/Cc	60/143-	\ \	<b>†</b>	<b></b>	<b>^</b>	\ \ \	1	1	Î	15/143	_	€0/143	_	15/143-	1	1	£41/09	60/143 15/143	641/09
	·																		<i>3</i> 7.
													A.44						
								_					1000						

TABLE II (Continued)

Compounds of Base Polymers 1785-47A, 1829-20, and 1869-34 (parts by weight and test results)

₹₽₺ <u>. 34~₹</u>	<b>1</b>	86.8	180	41	
20-14 34-A 34-B   24-C   24-D   5-E   34-F   34-G   24-H   24-I   Ret. 34-I   Ret. 34-I	*Milled and heated 66 Hrs. @ 930c.	6.06	180	777	
34-1		66.8	200	1,2	
34-1		8.99	190	777	
24-H		49.2	120	52	
34-6		56.2	110	50	
2 34-F 3 -75/93 -75/104 -75/121 16/143		8.99	100	55	
(A)		66.8	120	67	
34 <u>-D</u> 16/143		47.5	14.0	53	
34 <u>-c</u>		59.8	130	87	
34-8   75/93   75/93   75/143   75/143		54-4	150	54	
34-A		54.4	170	53	
70-17		62.3	180	45	
20-13		7.67	200	45	
20-12		76.5	200	777	
20-11	*Milled and heated 72 Hrs. @82°C. Bin aged 2.5 wks.	21.1	150	53	
47812 47816 47817 20-11 20-13 20-13 .5/93	*Milled *Milled and and heated heated heated [16 Hrs. 72 Hrs. © 100°C, © 82°C.]  Bin aged aged aged 2.5 wks.	35.2	300	77	
<u>47A16</u>		28.1	350	39	
47A12		56.3	310	36	
Batch Number Oven cure, Hr/C	Special Treatment	Tensile, Kg/cm <sup>2</sup>	Ult. Elongation, %	Shore A, Durometer	

TABLE II (Continued)

Compounds of Base Polymers 1785-474, 1829-20, and 1869-34 (parts by weight and test results)

Batch Number	0-75   N-75   W-75   7-76   7-76	34-L	34-M	N-75		34-P	34.40	34-R	34.5	34-T	34-10	37V	34-W	34-X	34-Y	34-Z	24-AA	34-AB	34-AC	34-P 34-9 34-8 34-8 34-5 34-1 34-0 34-V 34-W 34-X 34-X 34-X 34-Z 34-AA 34-AB 34-AB 34-AC RECUSAAC
Polymer 1869-34	100	100	100	3	8	100	100	100	100	100	35	130	100	100	100	100	100	100	81	100
Maglite D	5	5	5	5	*5	* 2	+5	-5	5*	<b>*</b> 9	<b>*</b> 5	<b>*</b> 9	5*	5*	5*	;	*5	2*	*5	*5
Dyphos		8				5	£	5*	5*		5*		5*	5*	5*	5*	*5	*5	5*	5*
2,6 diaminopyridine	5			2				2												
Chemlink 30									1.5	1.5		1.5								
Hisil 233	18		50	95	*02	-	28*	30*	30*							8				
Burgess OPE										100*	100*	100*	100*							
Mistron Vapor														30*		35-				
Aerosil 200															30*		23*		18	18
Nulok 321-SP																		80	*O77	-07
Silane A172	el		-	П	1	-1	1	1.5	1					1	1	1	1		-1	1
Dicup 40C	7	77	7	9						7.5										
Vulcup 40 KE					7	7	4	5	5		_1	7.5	7.5	9	77	5	77	5	5	5
Press cure, Min/°C	60/143	<u></u>	Ť		1	1	1		$\uparrow$	15/16	15/160	1	-60/143	1				→20/160-	1	$\uparrow$
Oven cure, Hr/C	1.5/100	$\uparrow$			1/100	力				24/100→	↑ Q	1	1/100		$\uparrow$	16/1	00 1/10	16/100 1/100 1/100	$\uparrow$	18/100
	_					_		_	-	_	_	_					_			

TABLE II (Continued)

Compounds of Base Polymers 1785-47A, 1829-20, and 1869-34 (parts by weight and test results)

					_
34-P   34-9   34-R   34-S   24-T   24-U   34-V   34-W   34-X   34-X   34-Z   34-AA   34-AB   34-AC   RDt. 34AC		112.5	8	85	
34-AC		105.5	6	85	
34-AB		87.9	88	72	
34-AA		91.4	160	75	
<del>2-1</del> 2		84.4	0.11	77	
34-X		91.4	180	90	
34-X		42.2	120	775	
M-78		19.7	150	58	
34-1	Milled & Bin Aged 24 Hrs.	15.5	180	90	
34-10	<b>\</b>	17.6	150	74	
34-T	Willed & Bin Aged 24 Hrs.	44.3	120	62	
इन्ह		34.4	150	72	
34-R	<u> </u>	28.1	330	45	
877	<u> </u>	91.4	160	3	·
34-P	\\ 	91.4	150	72	
340	*Oven dried @ 127°C	80.9	800	73	
34-K 34-L 34-K 34-N 34-0		Stiff	300 able able		
34-14		Stiff	& Fri- able		
34-I		17.6	38	25	
34-K		35.2	>500	077	
Batch Number	Special Treatment	Tensile, Fg/cm	Ult. Elongation, %	Shore A. Durometer	
æ	Spec	Tens	nt.	Shor	

TABLE II (Continued)

Compounds of Base Polymers 1785-47A, 1829-20, and 1869-34 (parts by weight and test results)

Batch Number	34AU JAAE	3ME	34AF   34AG	34AG	34AH	Rpt. 34AH   34AI   34AJ   34AK	34AI	34A.	1 34A		34AL   34AM	34AN	34.40	J.A.D.	37.40	TALE SALE DAY		1111	
Polymer 1869-34	100	100	100	188	100	100	18	138	3		100	8	1 5	5	-	·		<u>.</u>	
Maglite D	2*	*5	*10	π. *_	*.	*5	2*		╛.	╛.	1.	1.	1.		3	3 3	3 3	5	3
Dyphos	*,	5*	*	*	†*5	*5	*5								1	*	7	2	*
Burgess OPE		*			T		v.					$\prod$				: +	*	ţ,	*
ISAF Black	25*		*07						1	$\downarrow$	$\perp$			+	+	$\dagger$	+	1	
Nulok 321 SP							*07	*07	* * *	*07						+	+	+	$\top$
Burgess KE												1001		+	$\dagger$	+	+		1
Aerosil 200			15*	T					$\perp$		1		3		+	1	+	+	
Apropri 200		†	1	1	1						9				3Q*	30.	8		
SELUSIT NO		1	1										20*			+		$\vdash$	$\dagger$
Aerosil R972				23*	28*	28*	23*	30*	30*	* 23*				+	-	+	$\dagger$	$\dagger$	$\dagger$
Silene EF				-	-									*	+	+	+	†	1
QUSO G32				T									+	3	+	+	+	+	1
QUSO WR82		†-	1		T									+		+	+	25*	25.
Carbowax 4000													+	+	+	$\dagger$	-	1	+

TABLE II (Continued)

Compounds of Base Polymers 1785-47A, 1829-20, and 1869-34 (parts by weight and test results)

													_
				L	1		_	-			_		
34A[	,I	$oldsymbol{\perp}$		L	7					8.99	800	20	
34AT					14	20/160				63.3	120	53	
3445	н			3		60/143	1	Ī	1	45.7	180	83	
34AR	1		7			Ť	1	1		12.2	100	88	
3449	٦	77		-	$\dagger$	#	1/100	1	<del> </del>	7.67	110	88	
JAP .	н		-		7	$\dagger$	-16/10	١,	<del>                                     </del>	70.3	100	9/	
24AN   24AC   24AC   24AC   24AC   24AC   24AC	1	-	$\vdash$		17	╫	#	1		73.8 7	8	778	
- F		$\vdash$	-	<del> </del>	7	╁	╫	1	-	56.2 7	18	69	
34AM   34	<del></del>	-	-	╁	7	150	1/100		_	84.4 56	140	87	
	٦		_	-	3	25/171120/160		L	_	i	88	75	
X 3.AL		_	_	-	5	<b>≯</b> 25/	<b> </b>	Ľ		6.76 5	L		
I 34AK	1	_	_	_	5		$\prod_{k}$	Ľ		112.5 105.5	%	62	
34AJ			L	L	L			Ľ			75	85	
34AI	1				5	20/160	18/100	/		116.0	88	₩	
Rpt. 34AH	1				7	1		,		5	180	719	
							8-			105.5	Ã		
34AD   34AE   34AG   34AH	H				7	$ \uparrow$	16/100 40/100 16/00 16/100 18/100	1		4.86	170	70	
34AG	Т				7	60/143	5 16/1¢	1		87.9	180	95	
34AF				_	ر. ا	1	16/10	4		38.7	200	78	
34AE					5	<del>                                      </del>	<u>φ1/0</u>			21.1	0 0 0 7	58	
QW.		-			5	20/160	0,100	*Oven	dried (	28.1	900 000 000 000 000 000 000 000 000 000	45	
				ide	-		12	¥	₽ <b>©</b>			-	
e.i		,	,	Perox	ì	in/c	20/	ment	ď	, E	on, %	eter	
Number	1172	101.0	1300	Buty1	40KE	mre, n	re, m	Treat		, Kg/c	ongati	Duron	
Batch Number	Silane A172	Luperco 101XL	Luperco 130XL	Di - t-Butyl Peroxide	Vulcup 40KE	Press Cure, min/°C	Oven Cure, hr./°C	Special Treatment		Tensile, Kg/cm	Ult. Elongation, %	Shore A Durometer	
щ	ß	ដ	ដ	E	K	五	16	ſΩ	•	ı#	<b> </b> 5	8	

TABLE II, Cont.

5°         7°         7°         7°         7°         7°         7°         7°         7°<	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	3 3 5 1	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	35 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	80* 5 7.5* 1	5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5
5         6         7	5. 5. 5. 35. 15. 25. 35. 35. 35. 35. 35. 35. 35. 35. 35. 3	5 - 5 - 1 - 1 - 5 - 1 - 1	5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5
54         54         54         54         54         54         56         56         56         60         40*	25 35 15 25 35 25 35 25 35 25 35 25 35 25 35 25 35 25 35 25 35 25 25 25 25 25 25 25 25 25 25 25 25 25	30.4	5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5* 5
35*         15*         .30*         50*         40* <td>35 15 25 35 35 4 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5</td> <td><del>┞╼╁</del><del>╒╼╂╍╂╍╂╼</del>╇╼<del>┽</del>╾┼╾┼╴┼┾</td> <td>15* 40* 40* 1 1.5 1.5</td>	35 15 25 35 35 4 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	<del>┞╼╁</del> <del>╒╼╂╍╂╍╂╼</del> ╇╼ <del>┽</del> ╾┼╾┼╴┼┾	15* 40* 40* 1 1.5 1.5
35*         15*          50*         40* <td>35* 15* 25* 35* 1 1 1</td> <td><del>┟</del><del>╒╺╃╍╃╍╃</del>╼╃╼┼╾┼┾</td> <td>15* 40*</td>	35* 15* 25* 35* 1 1 1	<del>┟</del> <del>╒╺╃╍╃╍╃</del> ╼╃╼┼╾┼┾	15* 40*
354         15°         50°         40° <td>35 · 15 · 25 · 35 · 35 · 35 · 35 · 35 · 35 · 3</td> <td><del>┞┈┇┈┋┈┋┈</del>╃┈┽┈┼┈┼╴┼┼</td> <td>1 1.5 1.5</td>	35 · 15 · 25 · 35 · 35 · 35 · 35 · 35 · 35 · 3	<del>┞┈┇┈┋┈┋┈</del> ╃┈┽┈┼┈┼╴┼┼	1 1.5 1.5
1         25*         23*         23*         23*         23*         18*	25. 35.	<del>┠┈┝╍┠┈</del> ╅┈╅┈┼┈┼	1 1.5 1.5
18			1 1.5 1.5
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			1 1.5 1.5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 6 7 7 7 7	5 1 1	2	5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 6 64 68 59 71 70 60 78 73 84	φ <b>λ λ</b>	2	5 5 5
5         5	5	5	5 5 5
98.4 92.9 37.9 102.0 91.4 91.4 73.8 98.4 102.0 94.9 91.4 63.3 84. 75 80 64. 68 59 71. 70 60 78 78 78 78 78 78 78 78 78 78 78 78 78		4	
98.4         94.5         87.9         102.0         91.4         91.4         91.4         73.8         98.4         102.0         94.9         91.4         63.3         84           75         80         20         10         100         90         70         90         90         90         75         110           75         80         64         68         59         71         70         60         78         75         83         73			
98.4         92.9         97.9         102.0         91.4 <t< td=""><td>1</td><td></td><td></td></t<>	1		
98.4         94.5         97.9         102.0         91.4         91.4         73.8         98.4         102.0         94.9         91.4         63.3         84.           90         90         27.0         10         90         70         90         90         75         110           75         80         64         68         59         71         70         60         78         75         83         72			
98.4         92.9         87.9         102.0         91.4 <t< td=""><td>_</td><td></td><td>@127°C</td></t<>	_		@127°C
30         90         10         100         90         70         90         80         75         110           75         80         64         68         59         71         70         60         78         75         83         73	6.76	7-86	
75 80 64 68 59 71 70 60 78 75 83 72	96 96	8	250 100 110 130 80
	88	7.0	28 60 75 70 70

A STATE OF THE PARTY OF THE PAR

TABLE II, Continued

L'741 oneil	100 100 5* 15* 10*	77.3	
47-Alonsil   47-Alonsil A   1741 ones	100 5* 15* 20*	80.9 110 775	
	1 1 1 1 1	5 67 1100 54	
ਸ਼ਾਹਾਬ 27   ਨਹੁੰਸ   ਨਹੁੰਸ   ਤੁਸਾਂ   ਤੁਸਾਂ   ਤੁਸਾਂ   ਰਹੁੰਸ   ਰਹੁੰਸ	5*	180	
727	100	8 1 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	
72.7	25 5	7,110	
127	100	1 1.5 2 5 5 5 5 77.8 87.9 91.4 80 120 110	
577	100	1 8 2 1 1 1 2 1 1 1 2 1 1 2 1 1 2 1 1 2 1 1 2 1 1 2 1 1 2 1 1 2 1 1 2 1 1 2 1 1 2 1 1 1 2 1 1 2 1 1 2 1 1 1 2 1 1 1 2 1 1 1 2 1	
<u>27.7</u>	85 × 25	2.5	
27	100 5* 25*	2.0	
<u>az</u> 7	25, 25,	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	
		1 2 70° 5 110° 5	
NZT NZT	100 25.5 2.5		
WZ77	25* 25*	14.0 14.0 88	
12年	100 25 25 15	34,112	
XZ7	100 5* 25*	32 20 20 20 20 20 20 20 20 20 20 20 20 20	
£7.5	100 5* 25* 25*	2 1 50 4 6 7 6 7 6 7 6 7 6 7 6 7 6 7 6 7 6 7 6	
127	100 25* 25*	277 277 277 277 277 277 277	
[25]	100 * 5* * 25* 1.5	70.3	
127   327   127   127   127   327   327	100	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	
1277	100	5 5 5 20/150 > 1/121 -> *0ven dried > 120 120 120 120 120 120 120 120 120 120	
Ba 'h Wumber	Polymer 1785-47A Maglite D Acresil 200 Alon Silane A151 Silane A171	mir. log wir. log tracut tracut lon, % ometer	

TABLE II, Cont.

İ	*5	2*	2 *	2*	31 \$	3	100	100	9	188	100	2 130	81 ~	100	100
						<b>*</b> 9	*9	*9	*9	*9	*9				
	18*	25*	32*												
				25*	32*										
						30*	30*	30*	30*	30*	30*				
												5C+	8		
														50	8
	5	5	5	5	70	9	9	9				2	2	5	V.
									9	9	9				
Press cure, Min. /C	20/160 —	$\downarrow$	1		$\downarrow$	15/160 -	1				1	20/160	1	<b>\</b>	1
	1/121 —		<b>\</b>	<b>\</b>	$\downarrow$	24/100 -	\ \	$\downarrow$		<b>\</b>	1	1/100	1	1	1
	*oven dried @ 127°C					<b>1</b>	*Milled and Bin Aged 24 F Rm.T.	*Vac. oven 24 Kr. @ 80 G @ Bin Aged 24 Hr Rm.T.	*Oven dried @ 127°C	*Milled & Maged 24 Hr @ Fm T	1 *Vac- 1 oven 24 Hr 7 980 C Bin Aged 24 Hr 9Rm 7	*Oven +	<b>1</b>	<b>A</b>	1
	31.6	63.2	4.1	63.3	63.3	6•08	80.9	91.4	73.8		73.8	35.1	28.1	33.7	28.1
Ult. Elongation, %	8	150	8	150	011	210	210	250	260	350	300	150	500	200	200
	09	79	46	63	2/8	65	65	19	65	61	58	52	58	12	56

TABLE II, Cont.

47224	L	-			*.	28*		1	+	1	1				80.9	200	63		21.1	120	72
47223	*5			*5		28*		1	4	<b>\</b>	,				35.2	180	55		7.7	98	3
100	5*		5*			29*			4		. /				64.0	210	61		24.6	100	70
47ZZ1 100	5*	5*				28*		1	4	1	, /				77.3	200	62		35.1	100	72
0ZZZ77	*5					28*		1	7		/	ì			73.1	200	62		26.7	011	69
47XX4 100	5*				1*	23*	40*	1	5	1					84.4	100	70		40.1	50	92
47XX3	5*			5*		23*	*07	1	5	1	. ,				47.1	100	3		21.1	90	72
7XXZ7	5*		5*			23*	*O*	1	5		. ,				73.8	100	89		42.2	50	77
47XX1 100	*5	5*				23*	*0†	1	5		. ,				87.9	100	72		45.7	50	92
4.7XX0	*5					23*	*O7	1	5	20/1/02	20,5		*Oven dried	6, 127°C	85.8	100	70		12.1	50	712
Batch No. Polymer 1785-47A	Maglite D	Dyphos	Dythal XL	Leadstar	SAF Black	Aerosil R972	Wulik 321 SP	Silane Al72	Vulcup 40 KE	Press cure. Win. / C	,	Oven cure, Hr./ C	Special Treatment		Tensile, Kg/cm <sup>2</sup>	Ult. Elongation, %	Shore A Durometer	After Aging 70 Hrs. @ 150°C		Ult. Elongation, %	Share A Durameter

TABLE III
TABLE Polymer 1931\_07

						(Parts by	(Parts by Weight and Test Results	tht and	and Test R	Results)									
Batch No.	A LO	OIB	010	OLD	OLE	OLF	010	нто	011	01.0	Olk	OIL	MIO	ाम	OIP	010	02A	CZB	020
Polymer 1931-07	٤	80	100	138	81	201	100	100	100	100	100	100	100	188	100	81	183	100	100
Mamosium oxide	9	9	9							, r			5		2		5	5	
Perhon	,	, ,,						1		5	5						1		
Zine oxide			5				3	1	5			5	5			2			5
Caloism bradaovide	<b> </b>			9	9	9	9	5			5	5							
Calcium nyuroxiue				Ì										9			3		
Cadimin Stearare																		3	
Stearic acid							1												
Dioctvladipate	<u> </u>						10												
Aerosil R972				0†				20		20	20	25							
Nulok 321 SP								07		07	07	35							
Hisil 233					04	30	30		20										
Mistron vapor									20										
TSAF black													07	07	30	30	1		
MPC black																	30	30	
HAF black																			30
Silone A 172						1		1.5	1.5	1.5	1.5	1.5		1.5				2	2
Carbonax 4000									1.5										
Vileno 40 KB	9	9	9	9	9	9	9	5	5	5	3	5	5	5	5	2	2	1	7
Press cure. Min./C	1	25/160 →		\rightarrow{1}{\right	Image: Control of the control of the	<u></u>	$\downarrow$	1	1	1	1	1	20/1	20/160		1	$\uparrow$	$\downarrow$	<b>\</b>
Oven cure, Hr./'C		1/121	1		1	$\downarrow$	$\int$	1	$\downarrow$	1	1	1	1		$\downarrow$	$\overline{\mathbf{A}}$	1	<b></b>	<b>1</b>
Special Treatment																			
				į	1	9	t	- 60	. 6. 7	5	5	777 3	N.O.	, i	7.7.	Ş	27.6	5,42	NO.
Tensile, Kz/cm Ult. Elongation, %	77	150	170	)     	99 98	138	a Shorte		8,5	186	87	185	Cure	Curre	250	Cure	87	82	Cure
Shore A Durometer	<del></del>	ŏ		17)	2	3		0	2	2	3	5		_ •			1		

TABLE III, Cont.

NAME OF THE PERSON OF THE PERS

Batch No.	02D	OZE	OZF	02G	02H	021	223	02K	07A	O//B	O/BE	# / S	2	3	5	25	5	1/0	1
Polymer 1931-07	100	001	100	<b>*</b> 05	1001	100	00T	300	100	100	55	2	æ	100	188	81	8	100	Ì
Royalene 400				50*							50	30	20						1
Magnesium oxide		5			9	5	5	. 5.	5	5*	5	5	5	5	5	5	5	5	1
Dyphos						5	5		5.	5*	5	5	5	5	5	3	5	2	•
Cadmium stearate			5																1
Zinc stearate				5*															١
Calcium hydroxide	4							5											
Agerite Resin D			25	1.5*															١
Chemlink 30					1.5														1
HAF black	30	30	30	30*	07														١
Mulok 321 SP						04	07	70	07	<b>*</b> 07	04	07	07			20	30	70	١
Aerosil R972					20	15	20	20		23*	23	23	23	1,0	T <sup>†</sup> C	07	30	20	1
Aerosil 200									18							1			Ì
Silane Al72	2	2				1	1	7.5	1	1	г	п	1						1
Carbonax 4000												2	2		2	٠,	~	т.	-
Vulcup 40 KE	7	7	7	7	9	5	5	5	5	5	5	5	5	5	5	5	5	2	Ì
Press cure, Min./°C	20/160 →	<b>1</b>	<b>\</b>	1	25/160	₹ 091/02	\ \ \ \	\ \	1	<b>\</b>	+	1	1	1	1	25/160 ▶	39	<b>↑</b>	
Owen cure, Ar./°C		2/121 →	1/121	1	<b>↑</b>	1/100 →	<del> </del>	1	1	1	1	1	1		1	+	1	<b>↑</b>	
Special Treatment				*Bin Aged 2 Wks @ Rm.T.	T.					*Oven dried @ 127°C	ပ								
Tensile, Kg/cm <sup>2</sup>	No	No	No	63.3	No	84.4	84.4	84.4	73.8	98.4	59.8	36.9	35.2	56.2	7	71.7	87.9	73.8	
8	Cure	Cure	Sure	650	ē.	110	8	110	20	%	007	160	360	100	200	17,0	120	120	- 1
Share A Durameter				50		65	02	29	202	62	55	63	8	39	62	20	3	55	_
•						•			•										

TAHLE III, Cont.

Batch No.	07.3	οχκ	27,0	E	ğ	Š
Polymer 1931-07	100	100	100	188	18	18
Magnesium oxide	5	9	9	9	5	
Dyrphos	3				5	5
Cadmium stearate	3					3
Aerosil Ryr	30		30	30	23	23
Hisil 233		30				
Mistron Vapor				50		
Nulok 321 SP	30				07	07
Silane A172			1	1	7	1
Carbowax 4000	2	2	7	7		
Vulcup 40 KB	5	7	4	5	5	5
Press cure, Min./ C	25/160	20/160	1	<b>\</b>	\frac{1}{2}	1
Oven cure, Hr./ C	001/ر	1/1.1	$\downarrow$	<b>\</b>	1/100	1
	·					
Tensile, Kg/cm <sup>2</sup>	12.2	42.2	1.5.7	49.2	91.4	73.8
Ult. Elongation, %	220	220	300	130	8	14,0
Shore A Durometer	53	22	59	\$	73	62

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